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VentureStarTM

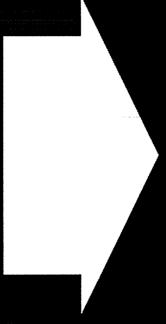


Developed under NASA MSFC Cooperative Agreement NCC8-115 dated July 2, 1996.

06/13/98

Philosophy of VentureStart^mOperatio

- Operability in Vehicle and Ground System
- Rapid Turnaround Cycles
- -Increase Launch Rate
- High System Reliability/Availability
- Reduce Operations Personnel



Low Cost Access to Space

Philosophy of *VentureStar™* Operatio

- Design Operability into the Vehicle
- Single Stage to Orbit
- Common Propellant Based System
- MPS,RCS, Power Generation
- Standard Vehicle/Payload Interfaces
- Vehicle Health Monitoring
- Metallic TPS
- Minimize Fluid Requirements
- No Hypergols
- No Distributed Hydraulics





Philosophy of VentureStar MOperation

Design Operability into the Ground System

- Runway - to - Pad - to - Orbit

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Descent & Landing

Minimizes Vehicle Transfers

Minimizes Umbilical Demates and Retests

Minimizes Duplicate Ground Support

Capabilities

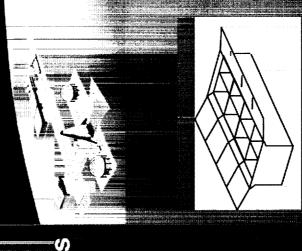


Independent Mission Module

Offline Hazardous Operations

Reduces Impact to Vehicle Processing

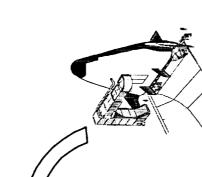
- Automate Ground Operations





Mission Operations

- Deploy ASAP Return first opportunity to Launch Site



Landing Operations

Launch Operations

- Vehicle Rotation
- Propellant Loading
 - Crew Ingress
- Late Stowage
- Vertical Launch



- Tow to Pad

Payload Processing (Offline)

- Safing



- Mission Module Removal
 - Vehicle Maintenance

- Mission Module Installation



Mission Module Interchange

- Payload Processing/ Installation - Mission Module Maintenance

- Vehicle Servicing

Operations Control Center

- Ground Operations
- Launch Operations
- Mission Operations

